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Abstract #1666



English

The latest from MOXIE

Essential subsystems of the Mars Oxygen ISRU Experiment (MOXIE) on NASA's Mars 2020 mission have passed their Critical Design Review, and the project is beginning a flight build stage that will culminate in delivery of flight hardware in late 2018. Also being delivered to the MOXIE Science Team is an Engineering Model, equivalent in form, fit, and function to the flight hardware, that will support Mars 2020 mission operations and subsequent extensions of the MOXIE flight demonstration (for example, tests of long duration operation). As a scale model of a future oxygen production facility on Mars, MOXIE will generate a minimum of 6 g/hr high purity oxygen from the Martian atmosphere in at least 15 separate runs, sampling different environmental conditions on Mars, in the 2.5 years following landing of the Mars 2020 rover in February, 2021. The presentation will review the MOXIE scroll pump for CO2 collection and compression, the solid oxide electrolysis system and its packaging for CO2 conversion to O2, the monitoring and control subsystem, and the expectations for operation on Mars.

French

No abstract title in French

No French resume

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N/A The MOXIE Team (UnknownTitle) MIT Haystack Observatory 4/26/2017 CIM | TPMS |

Profile of Dr. Michael Hecht



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Preferred Language: [Language not defined]

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Biographies

Biography submitted with the abstract

Michael Hecht is the Associate Director for Research Management at the MIT Haystack Observatory, and Principal Investigator for the MOXIE payload on NASA's Mars 2020 Rover mission. Previously, he was a Senior Research Scientist at JPL, where he served as PI and Instrument Manager for the MECA instrument on the Phoenix Mars Scout mission, a soil analysis suite for investigating wet chemistry, microscopy, and thermophysical properties, which was originally developed in support of the human exploration.

Biography in the user profile

Collaborators

Author(s) and Presenter(s)

Author(s):

Dr. Michael H Hecht

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[Unknown Title] MIT Haystack Observatory

N/A The MOXIE Team [Unknown Title] MIT Haystack Observatory

Presenter(s):

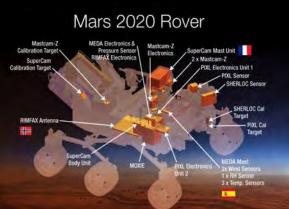
Dr. Michael H Hecht [Unknown Title] MIT Haystack Observatory





The latest from MOXIE

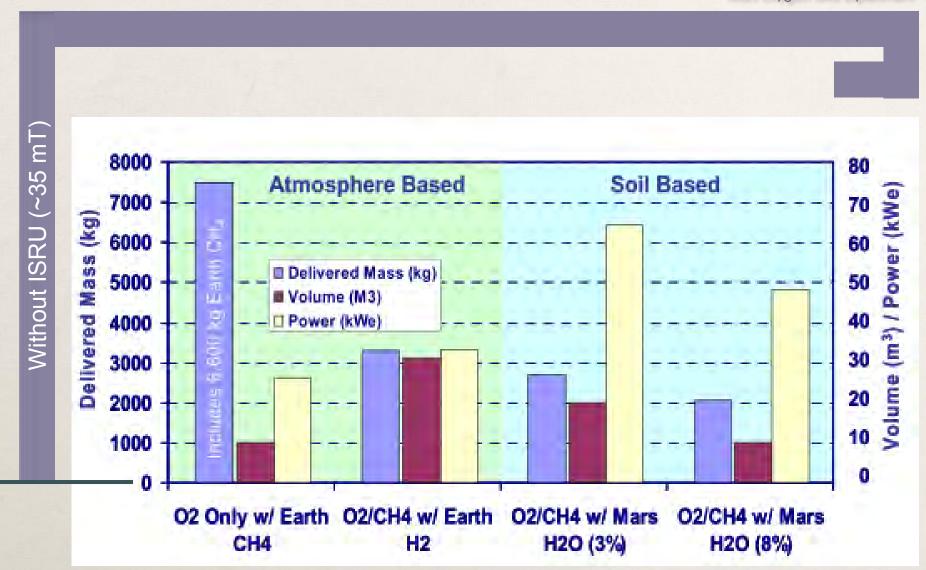
Michael Hecht & The MOXIE Team





ISRU & the human Mars mission

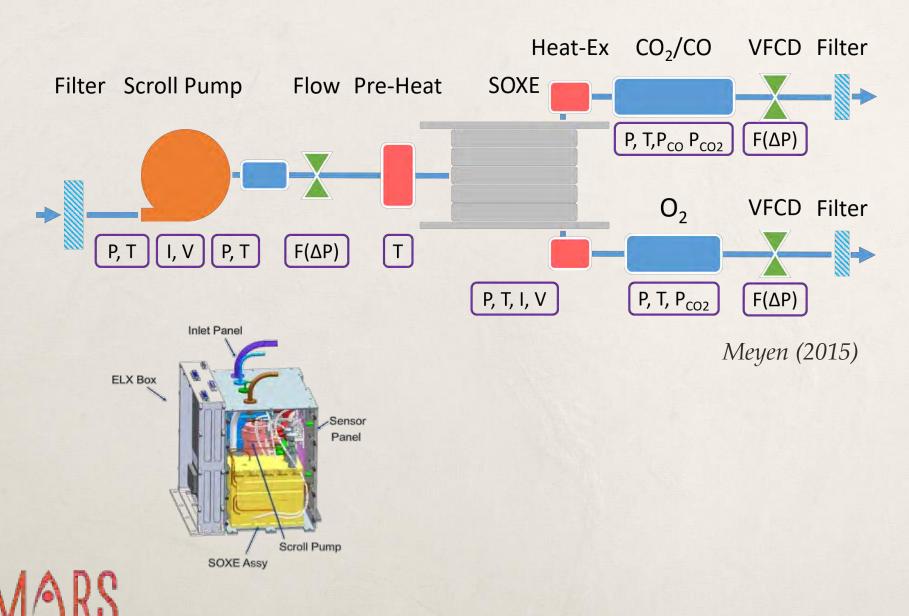






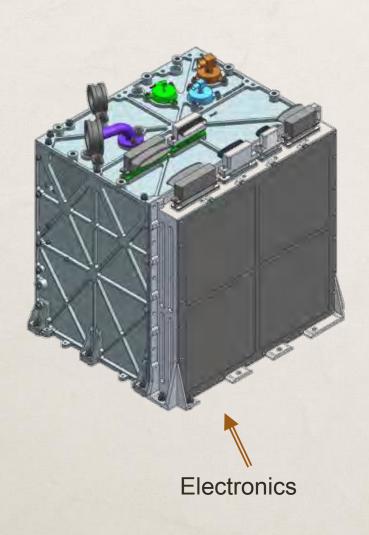
How does it work?

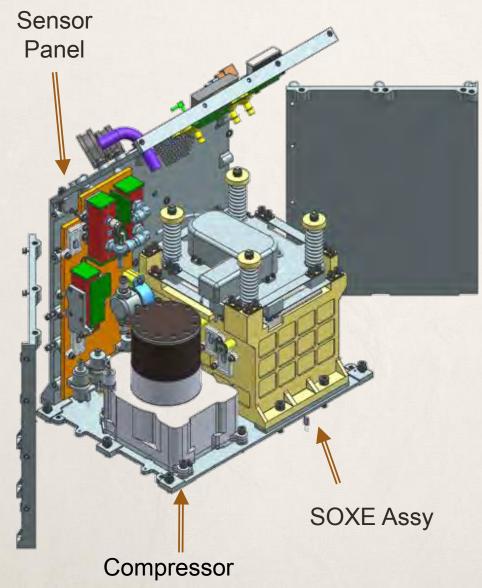




How does it all go together?

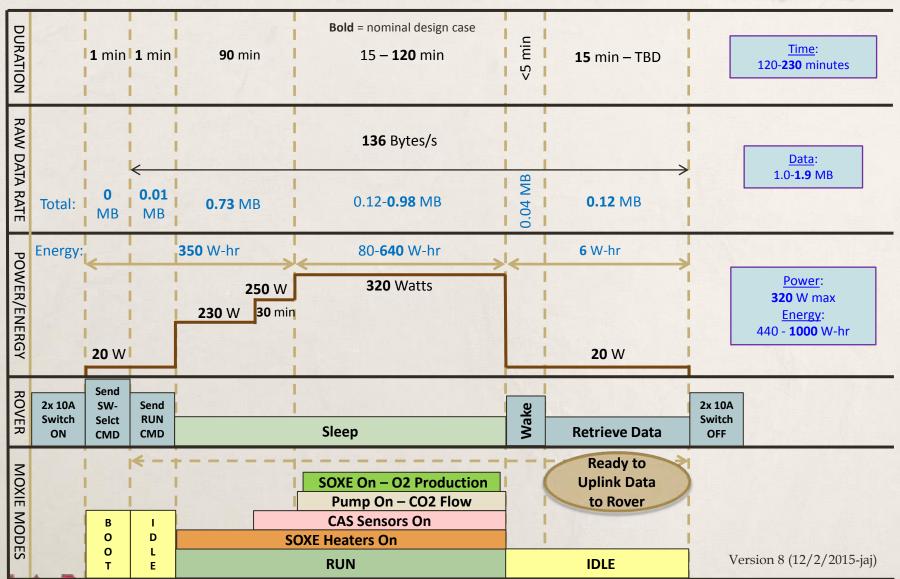






A day in the Life





Formal Requirements



Requirement	Goal	Threshold	
Oxygen Production Rate	8 g/hr at 5 Torr, 0°C.	6 g/hr at 5 Torr, 0°C.	
Oxygen Purity	99.6%	98%	
Number of Cycles	20	10	

System Performance

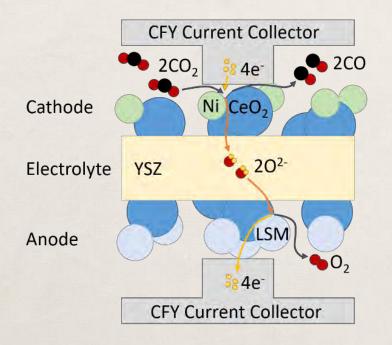
- O₂ production >1g/hr per cell (10 cells)
- Operational Cycle >45 cycles w/ no failures
- O₂ purity All recent stacks exceed 99.9%
- Limited by:
 - Inlet flow (pump capacity, gas density at landing site)
 - Available power (4A limit, equiv. to 12 g/hr)
 - SOXE capability (10 cells, 22.7 cm²/cell)

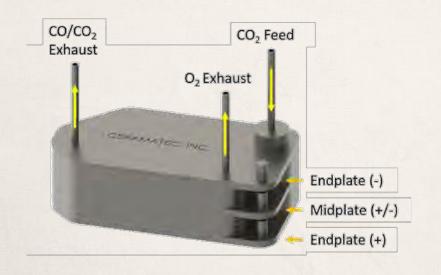


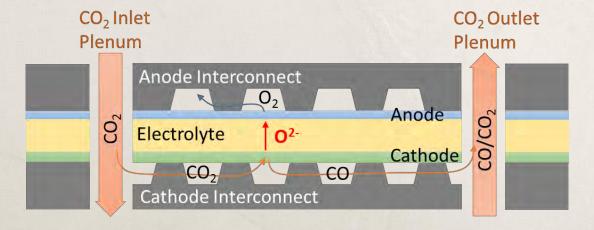
Making O₂ with SOXE









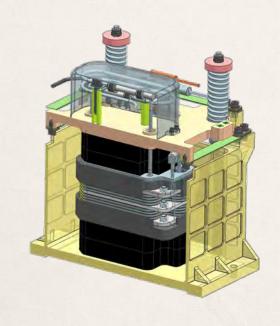


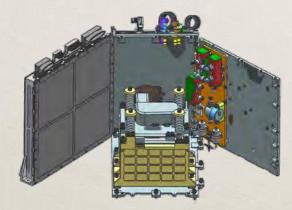


SOXE Assembly











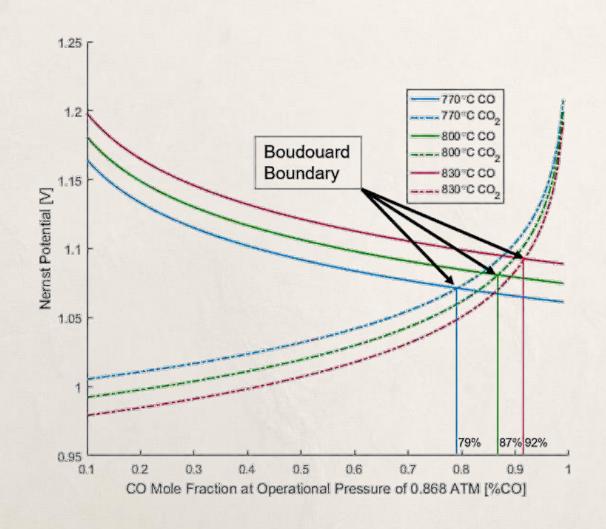


Boudouard and Ni Oxidation Limits





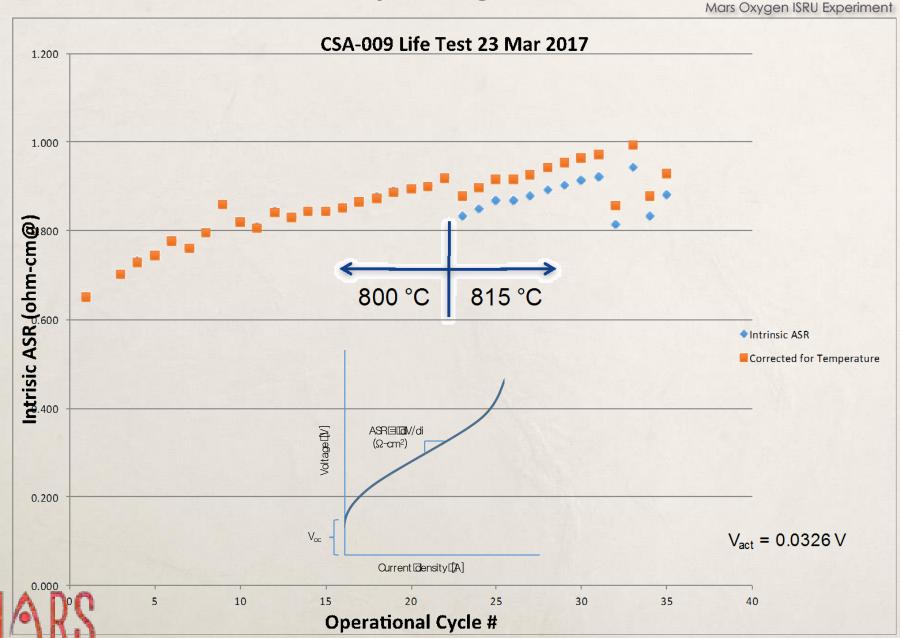






ASR resilience to cycling



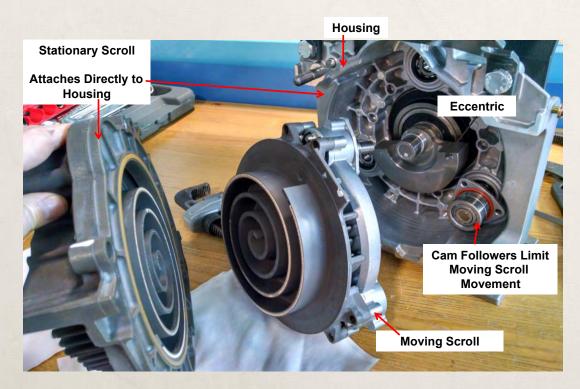


CO₂ compression strategy



- * Scroll pump chosen for real time compression to ~1 bar without intermediate storage.
- * Energy efficient, can be scaled at least 10-fold. Lifetime TBD.
- * Performance: 83g/hr for inlet gas P=7 Torr, T= 20°C.





MARS

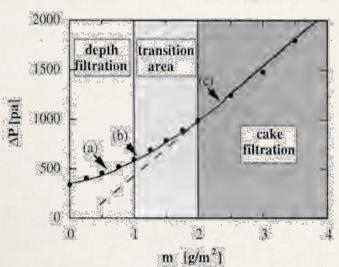
Dust concern retired



- * During operation (~100 hrs):
 - * Airflow a few cm/s
 - * Expect $\sim 50 \text{ mg dust}$, $< 0.1 \text{ g/m}^2$
- * Passive exposure (>10,000 hrs)
 - * Expect $10^{-6} \text{ g/m}^2\text{-s} \rightarrow -2 \text{ g/m}^2$
- * Larger particles
 - * Saltation, pebbles will be stopped by baffles

Clean	Passive	Active	

Dust exposure at Aarhus wind tunnel



- * Significant pressure drops begin at 1 g/m² for d_p =0.15 µm. (*Thomas et al.* 1999)
- ★ Cross section correction →
 expect no problems below ~10
 g/m²
- * Tests confirm no effect for comparable loading with Mars simulant (2-10 μm)
- * Slip factor provides up to x10 additional margin



State of the Project



- * All except electronics & some mechanical through CDR
 - * Extended SOXE life test now at ~50 cycles
 - * Integrated tests exploring safe operating conditions
- * Outstanding issues
 - * Electronics/software behind schedule
 - * Some unusual (COTS) sensor behavior
- * On track for flight delivery in November, 2018
- * Flight-like engineering model to be delivered to MIT for mission support and ongoing testing ~4/19







Extensibility

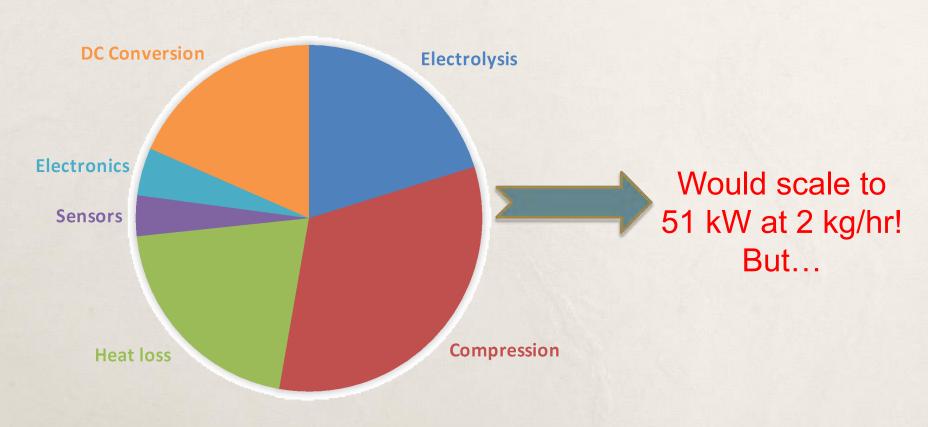
Power scaling Mass & Volume scaling



Where does the power go?









Power scaling assumptions



- * Scales with production rate R (2000/12 (g/hr) = x167):
 - * Electrolysis (including enthalpy) rigorously!
 - * Compression
- * Scales with # of modules (x6, 334 g/hr each, \sim sixty 10x10 cm cells)
 - * Sensors
 - * Electronics
- * Scales with surface area $R^{2/3}$ (x30)
 - * SOXE heat loss

* Expected improvements:

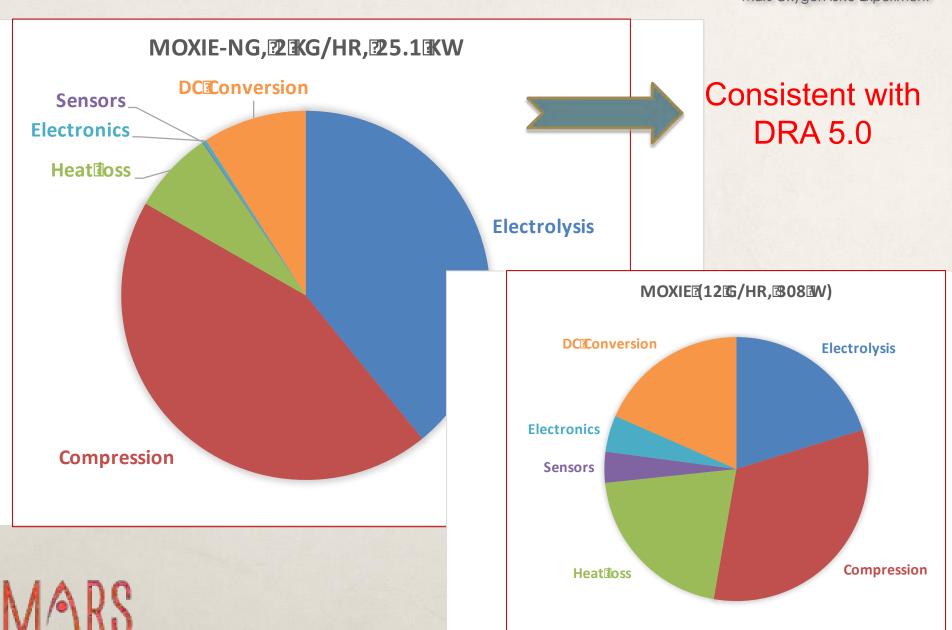
- * Compression power assumed to be 70% of scaled value
 - ★ Lower elevation (like MSL or VL2) gets you to 75%
 - * Reduce output pressure
 - * Increase utilization (more SOXE cells or CO₂ recovery)
- * Custom DC converters improve from 83% to 90% efficient
- * Gas pre-heat replaced by heat exchange with exhaust
- * Sensor panel captures heat from, e.g., pump body





Projected power usage

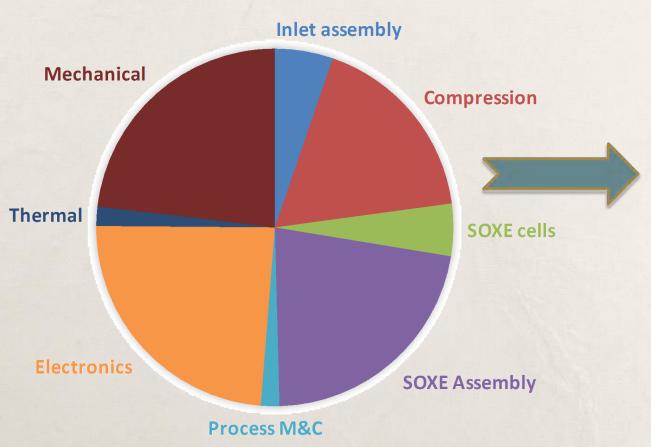




Where does the mass go?







Would scale to 2730 kg at 2 kg/hr!
But...



Mass scaling assumptions

Mars Oxygen ISRU Experiment

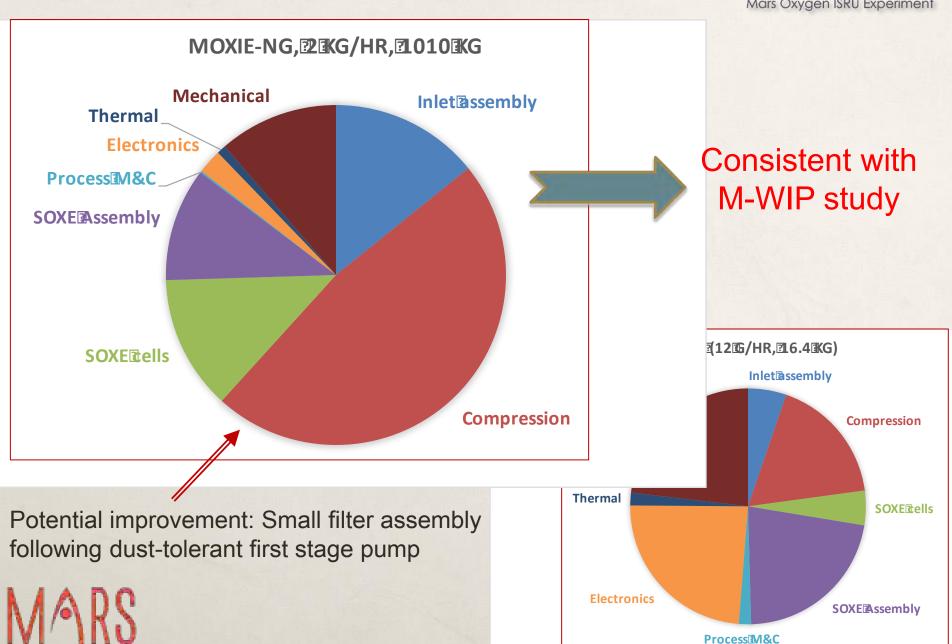
- \star Scales with production rate R (x167):
 - * SOXE cell mass
 - * Compressor mass
 - * Filter assembly
- * Scales with # of modules (x6)
 - * Sensors
 - * Electronics
- * Scales with surface area $R^{2/3}$ (x30)
 - * Thermal (insulation, etc.)
 - * Structure





Projected mass usage

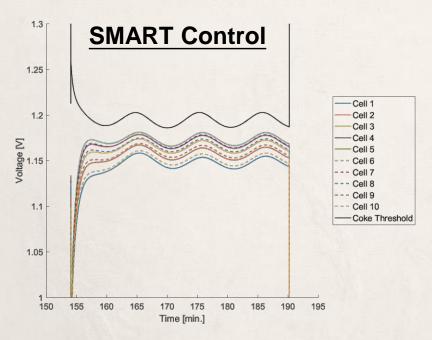




Other potential improvements



- * Higher temperature operation
- * Lower pressure operation
- * Higher utilization factor (low flow or CO₂ re-use)
- * Advanced controls and improved autonomy

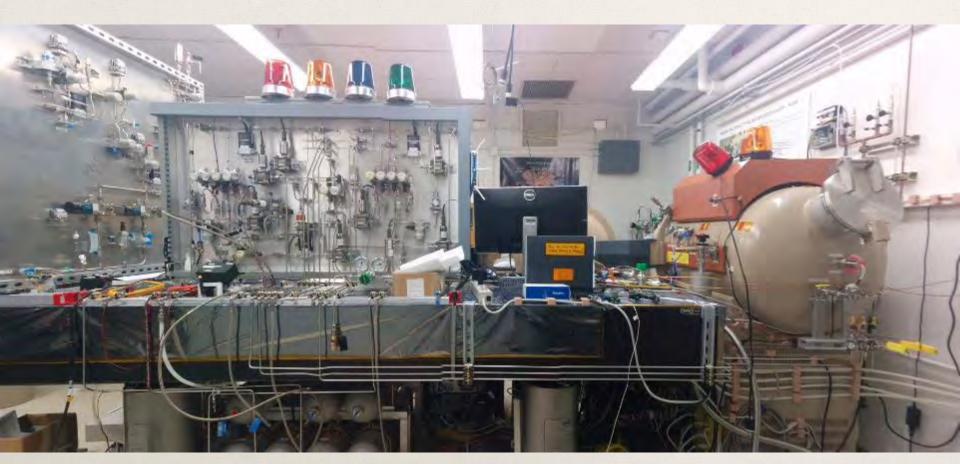


F. Meyen thesis





Special thanks to the amazing JPL Project Team!

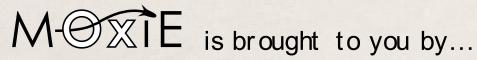




Sponsors and Partners



- Supported by HEO/AES, STMD/Tech Demos
- Mars 2020 Project managed by SMD





ISRU Technology, **NASA GRC** Aarhus wind tunnel



Massachusetts Institute of **Technology**



Jet[®] ropulsion **1** aboratory California Institute of Technology



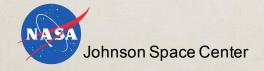






Imperial College London

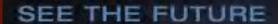






Congrats & Thanks, Forrest!





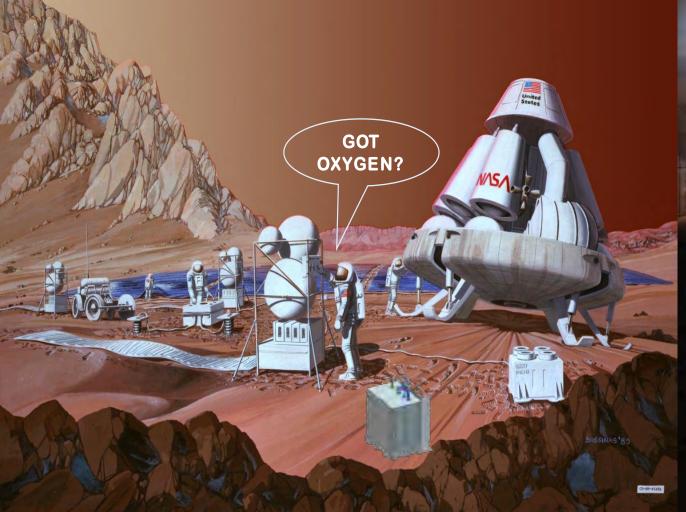


MARS 2020

SEEK PAST LIFE · COLLECT ROCK SAMPLES · PREPARE FOR HUMANS

#JOURNEYTOMARS:







Abstract

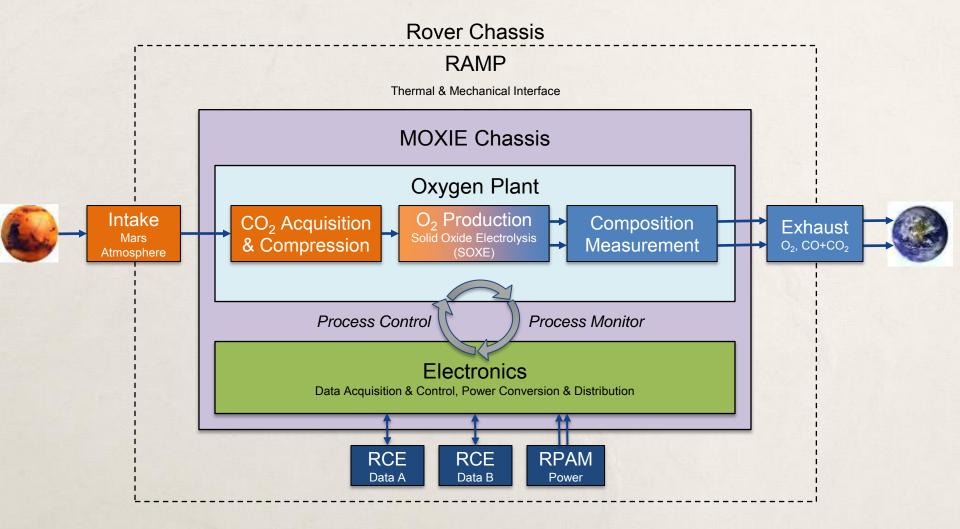


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What does MOXIE do?





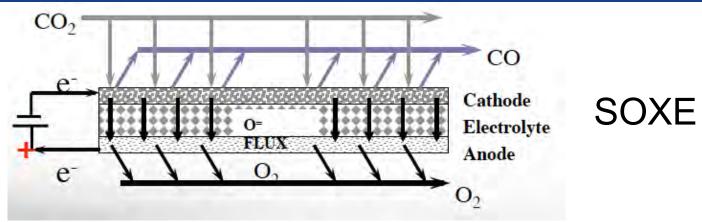


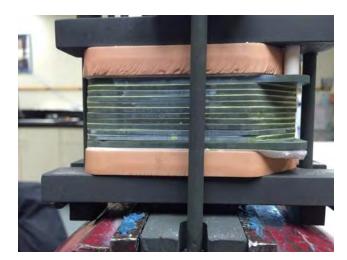
1:200 scale production

1:100 scale operating life

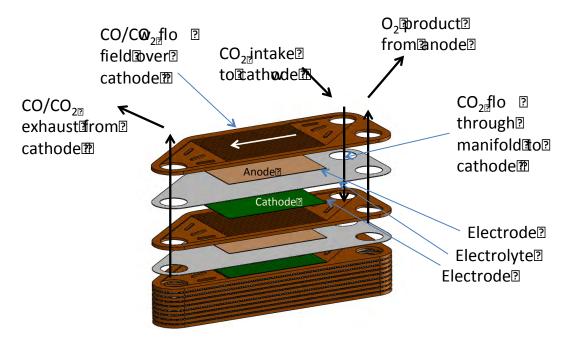








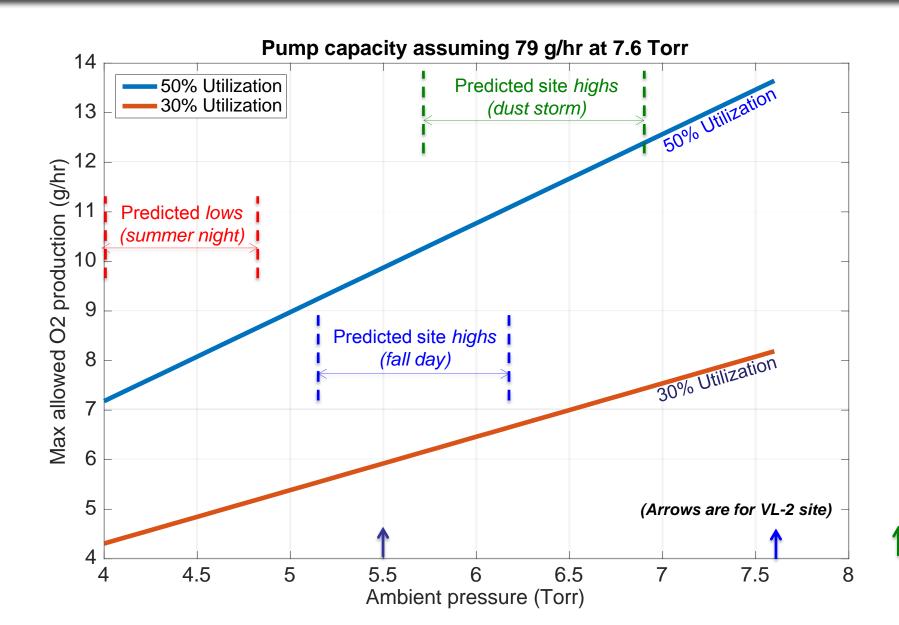
SOXE 11 cell stack fabricated by Ceramatec





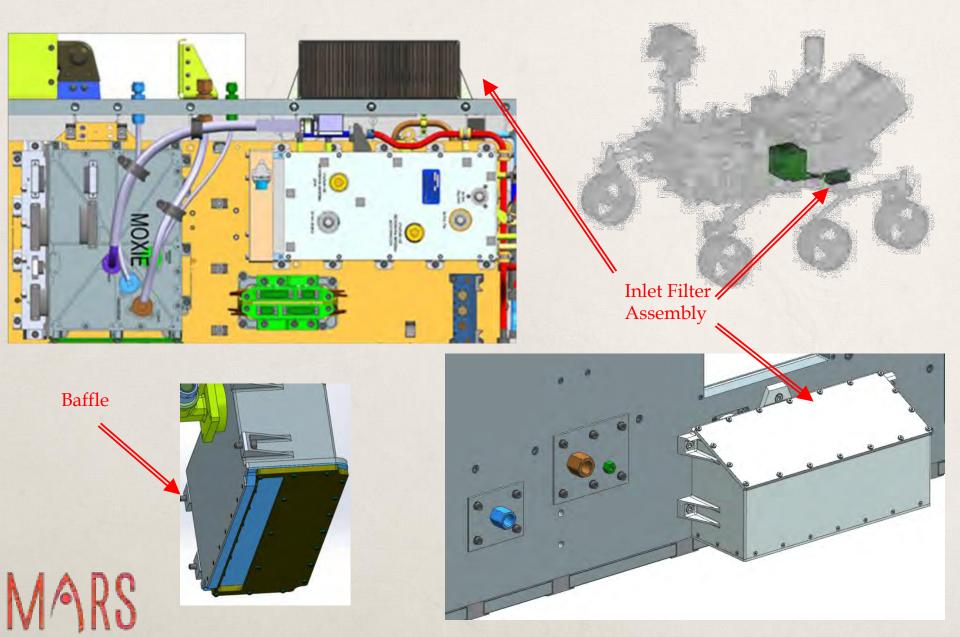
Pump capacity (at 20° C inlet)





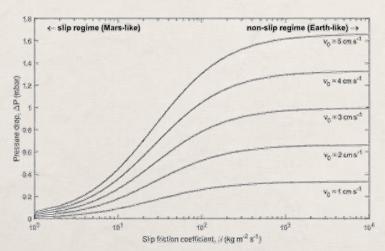
Inlet filter assembly& baffle



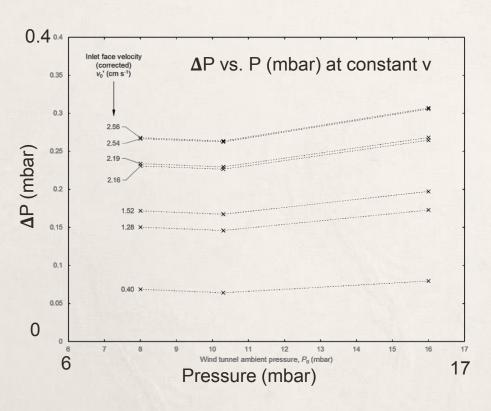


More on the Dust Story





Pressure drop of fluid with a viscosity 1.1x10⁻⁵ Pa-s, through a 0.4 mm long, 1 µm radius tube, using slip friction coefficients from 1 to 10⁴ and inlet velocities of 0-5 cm/s

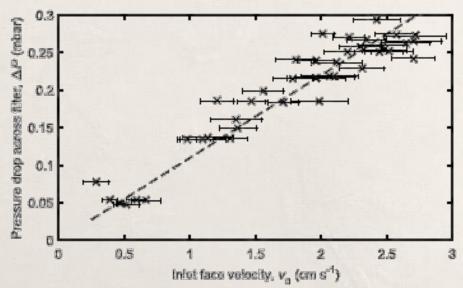


Measured result at Aarhus using laser Doppler anemometer to measure inlet velocity v.



Yet more on dust...





Pressure drop, ΔP , shown to scale with face velocity in Aarhus test

